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IN THE UNITED STATES PATENT AND TRADEMARK OFFICE

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Applicants: Steven A. Clark et al.  
Title: METAL ALLOY PRODUCT  
AND METHOD FOR  
PRODUCING SAME  
Appl. No. To be determined  
Filing Date: 09/07/2000  
Prior Appl. No.: 09/376,067

Prior Appl. Filing Date: 08/17/1999

Examiner: To be determined  
Art Unit: To be determined  
Attorney 43420/118  
Docket No.:

CERTIFICATE OF EXPRESS MAILING	
I hereby certify that the correspondence is being deposited with the United States Postal Service's "Express Mail Post Office To Addressee" service under 37 C.F.R. § 1.10 on the date indicated below and is addressed to: Assistant Commissioner for Patents, Washington, D.C. 20231.	
EL529676238US (Express Mail Label Number)	September 7, 2000 (Date of Deposit)
<p>Chris Escavallie (Printed Name) <i>Chris Escavallie</i> (Signature)</p>	

CONTINUING PATENT APPLICATION  
TRANSMITTAL LETTER

Assistant Commissioner for Patents  
Box PATENT APPLICATION  
Washington, D.C. 20231

Sir:

Transmitted herewith for filing under 37 C.F.R. § 1.53(b) is a:

Continuation     Division     Continuation-In-Part (CIP)

of the above-identified copending prior application in which no patenting, abandonment, or termination of proceedings has occurred. Priority to the above-identified prior application is hereby claimed under 35 U.S.C. § 120 for this continuing application. The entire disclosure of the above-identified prior application is considered as being part of the disclosure of the accompanying continuing application and is hereby incorporated by reference therein.

Enclosed are:

Specification, Claim(s), and Abstract (29 pages).

001.855148

Attorney Docket No. 43420/118

- Informal drawings (4 sheets, Figures 1-7).
- Copy of executed Declaration and Power of Attorney (from copending parent Application Serial No. 09/376,067, filed August 17, 1999, Attorney Docket No. 43420/117) (4 pages).
- Copy of executed Power of Attorney or Authorization of Agent by Assignee (from copending parent Application Serial No. 09/376,067, filed August 17, 1999, Attorney Docket No. 43420/117) (2 pages).
- Copy of executed Small Entity Statement (as filed in copending parent Application Serial No. 09/376,067, filed August 17, 1999, Attorney Docket No. 43420/117) (2 pages).
- Copending parent Application Serial No. 09/376,067, filed August 17, 1999, is assigned of record to Johnson Brass & Machine Foundry, Inc., as evidenced by assignments recorded in the U.S. Patent and Trademark Office at Reel/Frame 010503/0673-0678.
- Before calculating the filing fee, please cancel in this application original claims 2-16 of the parent application (Serial No. 09/376,067).

The filing fee is calculated below:

	Claims as Filed	Included in Basic Fee	Extra Claims	Rate	Fee Totals
<b>Basic Fee</b>				\$690.00	\$690.00
Total Claims:	24	- 20	= 4	x \$18.00	= \$72.00
Independents:	4	- 3	= 1	x \$78.00	= \$78.00
If any Multiple Dependent Claim(s) present:				+ \$260.00	= \$0.00
				SUBTOTAL:	= \$840.00
[ X ]	Small Entity Fees Apply (subtract ½ of above):				= \$420.00
				TOTAL FILING FEE:	= \$420.00

A check in the amount of \$420.00 to cover the filing fee is enclosed.

The Assistant Commissioner is hereby authorized to charge any additional fees which may be required regarding this application under 37 C.F.R. §§ 1.16-1.17, or credit any overpayment, to Deposit Account No. 06-1447. Should no proper payment be enclosed herewith, as by a check being in the wrong amount, unsigned, post-dated, otherwise improper or informal or even entirely missing, the Assistant Commissioner is authorized to charge the unpaid amount to Deposit Account No. 06-1447.

[ X ] Applicants respectfully request that the above-identified application be amended as follows:

**In the Specification:**

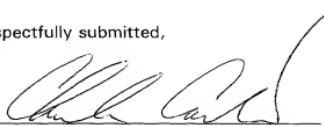
Please insert before the first line of the application:

"This application is a continuation of U.S. Application Serial No. 09/376,067, filed August 17, 1999, the disclosure of which is herein incorporated by reference."

Please direct all correspondence to the undersigned attorney or agent at the address indicated below.

Respectfully submitted,

By



Date Sept. 7, 2000

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**PATENT**

APPLICANT OR PATENTEE: Steven A. Clark et al. Docket No.:43420-117  
SERIAL OR PATENT NO.: 09/376,067  
FILED OR ISSUED: August 17, 1999  
FOR: METAL ALLOY PRODUCT AND METHOD FOR PRODUCING SAME

**VERIFIED STATEMENT (DECLARATION) CLAIMING  
SMALL ENTITY STATUS (37 CFR 1.9(f) and 1.27(c))  
SMALL BUSINESS CONCERN**

I hereby declare that I am:

[ ] the owner of the small business concern identified below.

[ X] an official of the small business concern empowered to act on behalf of the concern identified below.

NAME OF CONCERN: Johnson Brass & Machine Foundry, Inc.

ADDRESS OF CONCERN: 270 North Mill Street, Saukville, WI 53080

I hereby declare that the above identified small business concern qualifies as a small business concern as defined in 13 CFR 121.3-18, and reproduced in 37 CFR 1.9(d), for purposes of paying reduced fees under Section 41(a) and (b) of Title 35, United States Code, in that the number of employees of the concern, including those of its affiliates, does not exceed 500 persons. For purposes of this statement, (1) the number of employees of the business concern is the average over the previous fiscal year of the concern of the persons employed on a full-time, part-time or temporary basis during each of the pay periods of the fiscal year, and (2) concerns are affiliates of each other when either, directly or indirectly, one concern controls or has the power to control the other, or a third party or parties controls or has the power to control both.

I hereby declare that rights under contract or law have been conveyed to and remain with the small business concern identified above with regard to the invention, entitled:

METAL ALLOY PRODUCT AND METHOD FOR PRODUCING SAME

by inventor(s) Steven A. Clark et al.

described in:

[ ] the specification filed herewith  
[ X] application serial no. 09/376,067, filed August 17, 1999  
[ ] patent no. \_\_\_\_\_, issued \_\_\_\_\_

If the rights held by the above-identified small business concern are not exclusive, each individual, concern or organization having rights to the invention is listed below\* and no rights to the invention are held by any person, other than the inventor, who would not qualify as a

small business concern under 37 CFR 1.9(d) or a nonprofit organization under 37 CFR 1.9(e).  
NOTE: Separate verified statements are required from each named person, concern or organization having rights to the invention averring to their status as small entities. (37 CFR 1.27)

NAME: \_\_\_\_\_  
ADDRESS:  Individual  Small Business Concern  Nonprofit Organization

NAME: \_\_\_\_\_  
ADDRESS:  Individual  Small Business Concern  Nonprofit Organization

I acknowledge the duty to file, in this application or patent, notification of any change in status resulting in loss of entitlement to small entity status prior to paying, or at the time of paying, the earliest of the issue fee or any maintenance fee due after the date on which status as a small entity is no longer appropriate. (37 CFR 1.28(b))

I hereby declare that all statements made herein are of my own knowledge, are true and that all statements made on information and belief are believed to be true; and further, that these statements were made with the knowledge that willful false statements and the like so made are punishable by fine or imprisonment, or both, under Section 1001 of Title 18 of the United States Code, and that such willful false statements may jeopardize the validity of the application, any patent issuing thereon, or any patent to which this verified statement is directed.

NAME OF PERSON SIGNING: THOMAS H. PHEISTER  
TITLE OF PERSON OTHER THAN OWNER: VP-ADMINISTRATION  
SIGNATURE Thomas H. Pheister DATE Nov 4, 1999

**U.S. PATENT APPLICATION**  
**for**  
**METAL ALLOY PRODUCT AND**  
**METHOD FOR PRODUCING SAME**

Inventors:     Steven A. Clark  
                  Balathandan S. Pillai

## BACKGROUND OF THE INVENTION

It is well known to form an alloy body by static casting. Static casting includes pouring a molten alloy in a mold, solidifying. However, a problem with static casting is that the resulting alloy body is subject to impurities and high

5 porosity, both of which may reduce the strength of the alloy body.

It is also known to form an alloy body by a wrought method. Such wrought method includes heating an alloy to a temperature below its melting temperature, and striking the alloy to refine the grain size and reduce porosity. The resulting wrought alloy body has generally less porosity than an alloy body produced by static  
10 casting. However, the wrought method is often limited to the use of a small number of "standard" alloys, in addition to generally being more complicated and expensive than casting methods.

It is also known to form an alloy body by centrifugal casting. A centrifugally cast alloy body has generally less impurities and porosity than an alloy  
15 body produced by static casting. Aluminum pieces produced by centrifugal casting, however, still commonly have a significant amount of porosity and generally do not possess the overall strength and toughness properties that can be achieved with pieces created using wrought techniques.

To date, most centrifugally casting of aluminum alloys has been carried out  
20 using alloys with standard cast aluminum chemistries. Due to differences in alloy composition, pieces formed from alloys with standard cast aluminum chemistries are generally incompatible with wrought alloy bodies because the alloys formed by the wrought method and centrifugal casting generally have different physical and mechanical properties.

25

## SUMMARY OF THE INVENTION

The present invention relates to a method for producing cast alloy articles having high strength and/or toughness. The method includes providing a molten alloy, such as a molten aluminum alloy; centrifugally casting the molten alloy to  
30 form a cast body; and hot isostatically processing the cast body to form a hipped body. The hipped body may optionally be solution heat treated to form a heat

treated body, which may subsequently be precipitation hardened to further enhance the properties of the cast product as desired. The method allows the production of cast aluminum alloy articles having physical properties similar to those obtained for articles produced from corresponding aluminum alloy chemistries by wrought

5 techniques.

The present method can provide a centrifugally cast alloy body having physical and mechanical properties comparable to the physical and mechanical properties typically achieved with a wrought alloy. The method also can allow the production of a cast alloy body which has generally the same chemical composition

10 as a traditional wrought alloy. In many instances, the present method allows the production of a cast alloy body which, in addition to having generally the same chemical composition as a traditional wrought alloy, has many of the same physical and mechanical properties as pieces produced by traditional wrought techniques. This can permit the centrifugally cast alloy body to be coupled or welded to a piece

15 formed through common wrought methods. Various embodiments of the present method can permit one or more of the various advantages discussed above to be achieved. These and other advantages which may be achieved using the present method will be apparent to those skilled in the art upon review of the specification and the appended claims.

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#### BRIEF DESCRIPTION OF THE DRAWINGS

The invention will become more fully understood from the following detailed description, taken in conjunction with the accompanying figures, in which:

FIGURE 1 is a photomicrograph of a traditional wrought 6061-T651 alloy at a magnification of 100X;

25 FIGURE 2 is a photomicrograph of the alloy of FIGURE 1 at a magnification of 200X;

FIGURE 3 is a photomicrograph of a traditional centrifugally cast 6061-T6C alloy (i.e., not subjected to hipping) at a magnification of 100X;

30 FIGURE 4 is a photomicrograph of the alloy of FIGURE 3 at a magnification of 200X;

FIGURE 5 is a photomicrograph at 50X magnification of a 6061-T6C alloy produced by centrifugal casting and hot isostatic processing; and

FIGURE 6 is a photomicrograph of the alloy of FIGURE 5 at a magnification of 100X.

5 FIGURE 7 is a photomicrograph of the alloy of FIGURE 5 at a magnification of 200X.

#### DETAILED DESCRIPTION OF EMBODIMENTS

The present invention relates to a method for producing cast alloy bodies which includes providing a molten alloy, such as a molten aluminum alloy,

10 centrifugally casting the molten alloy to form a cast body; and hot isostatically processing the cast body to form a hipped body. The hipped body may optionally be solution heat treated to form a heat treated body, which may subsequently be precipitation hardened (also referred to herein as "artificially aged" or "heat aged") to further enhance the properties of the cast product as desired.

15 Melts may be prepared by heating metal, typically scrap or specially alloyed ingot, in a furnace. If the chemistry of the melt does not meet desired specifications, it may be re-alloyed as necessary with additions the requisite amounts of individual constituent elements. These additions are commonly made to the molten alloy ("melt") in the furnace. The metal which is used to form the melt,

20 whether scrap, alloyed ingot and the like or individual added constituent elements, is collectively referred to herein as "source metal." The chemistry of the alloy (i.e., the amounts of the individual constituent elements) is tightly controlled with respect to the amounts of both major and minor constituents. In some instances, it may be necessary to re-alloy the molten aluminum alloy with additions of minor

25 constituents, such as copper, silicon, magnesium, manganese, zinc or iron, as appropriate. The chemistry of a melt lot may be verified by computerized spectrochemical analysis prior to casting. Melt temperature will vary with the particular alloy composition and is established such that thorough mixing of the constituents is enabled as well as allowing the proper fluidity for the centrifugal cast

30 process. The temperature used should be low enough to minimize gas pickup, oxidation, and degradation of chemistry. For wrought aluminum alloys of the type

that are typically employed in the present method, melt temperatures of about 1,000 to 1,500°F (\_\_\_\_ to \_\_\_\_ °C) are common. For example, 6061 aluminum alloys are typically heated to about 1,400°F to form a melt.

There are various methods of heating metal alloys to form the molten alloy.

- 5 This can generally be accomplished by heating the source metal. According to a particularly suitable embodiment, the aluminum alloy is melted in an induction furnace, but other melting methods (e.g., gas furnace, convection melting, blast furnace, kiln, or molybdenum furnace) may be employed. Without intending to be limited by theory, it is believed that induction melting produces relatively low levels
- 10 oxides in the resulting melt as well as facilitating thorough mixing of the melted alloy.

The melted aluminum alloy is generally cast by pouring into a mold capable of being rotated at a relatively high speed (e.g., at least about 500 rpm). According to one common embodiment, the mold is in the shape of a hollow, walled cylinder

- 15 having an inside diameter about 4-18 inches less than an outside diameter. The interior and the exterior of the mold may be machined to an appropriate configuration. The mold inside diameter is typically machined to the appropriate configuration for the casting outside diameter allowing for any thermal contraction of the cast product which may occur during cooling.
- 20 The mold may be made of a variety of materials (e.g., steel, sand, graphite, and the like) having good dimensional stability and good heat transfer properties. The mold is generally made of steel, graphite, or other material capable of providing a high chill rate. From a cost/performance standpoint, mild steel and graphite are materials which are particularly suitable for use as mold materials in
- 25 conjunction with the present method. To facilitate removal of the cast piece, the mold may be coated with a protective insulating release agent, such as Permcoat or Centrificoat release agents. Molds made of graphite are quite suitable for use in the present method. Graphite molds having an inside diameter of about 10-45 inches are typically used in the present method. In most instances, the graphite mold is
- 30 encased in a larger mold mild steel mold. Although larger graphite molds may also be employed, it is quite common to centrifugally cast larger pieces using mild steel molds.

The melted alloy is poured into the mold, which may be pre-heated. Commonly, the metal is generally transferred directly from the melt furnace to a pouring ladle. The metal temperature is generally checked just prior to pouring. Metal is poured directly into the prepared centrifugal mold. The surface of the

5        melted alloy may be skimmed to substantially remove any floating impurities such as oxides. According to a suitable embodiment, about 4000 pounds or less of the melt is poured from a single lot of alloy heated and held in an induction furnace over a period of about up to about 8 hours.

The mold is generally rotated about a vertical axis during the pour. The

10      rotational speed of the mold develops a centrifugal force (e.g., G forces from about 30 to 130 G's). This produces an outward radial force applied to the mold as it is rotated. The centrifugal force is transferred to the molten alloy in the rotating mold through viscous effects. Rotation rates of at least about 500 rpm are commonly employed. The rotational rate is preferably sufficient to produce G forces of at least

15      about 60 to 70 G. The centrifugal force produces separation of impurities in the melted alloy based on differences in densities. As the melted alloy solidifies, impurities (e.g., oxides, dross, nonmetallic impurities and the like) that have a density generally less than the density of aluminum are forced toward the inside diameter of the casting. To a lesser extent, impurities that have a density generally

20      greater than the density of aluminum are generally forced to the outside diameter of the casting. Without intending to be limited by theory, it is believed that the centrifugal force reduces the amount of impurities and/or shrinkage defects (porosity) in the resulting centrifugally cast alloy body (relative to a statically cast body). The melted alloy solidifies until substantially no liquid metal remains in the

25      mold. The solidifying casting feeds progressively from the high pressure liquid metal inside the solidifying cylinder until no metal remains as the inside diameter becomes solid.

Unidirectional chilling of the metal may be assisted by applying a coolant, such as water, to the outside of the mold. During solidification of the molten alloy,

30      the temperature of the mold can drop from about 150 to 800°F over a period of about 10 to 120 minutes. The solidified alloy (i.e., the centrifgally cast body) may

be removed from the mold by overhead crane/hoist or by automatically ejection using conventional mechanical equipment.

The centrifugally cast body may be treated to produce a further reduction in shrinkage defects (porosity) by hot isostatic processing ("hipping") to form a hipped alloy body. Hipping is described in U.S. Patent No. 3,496,624 issued to Kerr et al., which is hereby incorporated by reference. Hipping includes elevating the temperature of the cast body in an autoclave to a temperature sufficient to achieve a solid state plastic condition and below the melting temperature of the alloy. For aluminum alloys, temperatures of at least about 850°F and more commonly about 900 to 950°F are employed. For example, with 6000 series aluminum alloys such as 6061 aluminum temperatures of about 925-985°F and, preferably from about 950-970°F, are typically employed in the hipping step. A high external pressure (e.g., via a pressurized gas such as argon or nitrogen) is applied such that a substantially equal force is exerted on each surface of the cast body ("isostatic pressure"). Pressures of at least about 10,000 psi are typically utilized. Preferably, isostatic pressures of about 10,000-20,000 psi and more preferably at about 14,000 psi are employed. Such temperature and pressure may be simultaneously applied for a period of more than 1 hour, typically for about 2-6 hours. Such temperature and pressure is intended to reduce the microporosity (microshrinkage defects) and densify the body by collapsing intergranular voids.

Elevated temperature develops a solid state plastic condition in a metal body (e.g., an aluminum alloy body). When heated to a temperature sufficient to achieve a solid state plastic condition while being subjected to very high external pressure, very small internal pores (referred to herein as "micropores" or "micro-shrinkage defects") can be forced to migrate out of the part. The behavior is analogous to squeezing a hollow lump of clay with your hand until it becomes a solid lump of clay. The temperature, pressure and time conditions employed to hip a particular alloyed product will depend on the alloy composition and to some extent, the size and geometry of the product. Different yet similar hipping procedures may be used as long as micro-porosity is substantially eliminated from the alloy material. In general, if the hipping process is carried out at a lower temperature (relatively), higher pressure and/or longer hipping times will be required to render the material

substantially free of micropores. As employed herein, substantially free of micropores means a material is substantially free of pores having a largest dimension which exceeds 0.0001 inch (0.1 mil).

Figures 1 and 2 are photomicrographs of a 6061-T651 alloy produced by the traditional wrought method. The expected elongated grain structure associated with wrought products is shown. The grains are generally about 3 times as long as they are wide. Their average size as measured by calipers off of the photograph is about 2300  $\mu$ inch by 900  $\mu$ inch (\_\_\_\_ mm by \_\_\_\_ mm). The elongated shape of the grain causes variations in directional properties. For example, rings cut from the plate shown in Figures 1 and 2 would possess dramatically different mechanical and physical properties in the longitudinal and transverse directions.

Figures 3 and 4 are photomicrographs of a centrifugally cast 6061-T6C alloy which has not been subjected to hipping. The material has a uniform and generally round grain structure is shown. The photomicrographs of Figures 3 and 4 also show small discontinuities representative of microporosity (i.e., micro-shrinkage defects). The defects show no specific shape and range in size from less than 1000  $\mu$ inch up to 4000  $\mu$ inch in size (\_\_\_\_ mm by \_\_\_\_ mm). These defects are believed to be responsible for the traditionally low elongation results from cast aluminum.

Figures 5 and 6 are photomicrographs of a centrifugally cast and hipped 6061-T6C alloy. This material has a uniform and generally round grain structure similar to the sample shown in of Figures 3 and 4. In contrast to the cast alloy material shown in Figures 3 and 4, the cast and hipped bodies (in Figures 5 and 6) show relatively few micro-shrinkage defects (i.e., the resulting body is substantially free of micro-shrinkage defects). The average grain size measures 3400  $\mu$ inch (\_\_\_\_ mm).

Aluminum alloy casting can generally be rendered substantially free of micropores by heating for a period of hours at a temperature of at least about 900°F (preferably about 925 to 990°F) while under an isostatic pressure of at least about 10 KSI. For example, micropores can be substantially removed from 6000 series aluminum alloy material (e.g., 6061 type aluminum) by placing the material into a

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hipping chamber, heating the material to about 960°F and holding the material at this temperature for about two hours while a pressure of about 14 to 16 KSI is applied.

The hipped body may be solution heat treated to further enhance its physical  
5 and/or mechanical properties. This is commonly carried out at a temperature in the range of about 900-1100°F, more preferably in the range of about 960-1000 °F for at least 1 hour, more preferable for about 6-8 hours. The hipped body may then be quenched with water, and then subsequently heated in a furnace at a temperature in the range of about 300-400°F, more preferably in the range of about 325-375°F for  
10 at least 1 hour, more preferably for about 7 to 10 hours. According to an alternative embodiment, the hipped body may undergo T6 heat treatment including solid solution treatment at a temperature in the range of about 900-1,000°F for about 2-8 hours, followed by water cooling or hot water cooling, and subsequent aging or age hardening at a temperature of about 325-375°C for about 4-15 hours.  
15 With respect to aging treatments, it should be noted that the hipped body may be subjected to any of the typical under-aging or over-aging treatments well known in the alloy casting arts, including natural aging. In addition, the aging treatment may include multiple aging steps, such as two or three aging steps. Also, stretching or its equivalent working may be used prior to or after part of any  
20 multiple aging steps. For two or more aging steps, the first step may include aging at a relatively high temperature followed by a lower temperature or vice versa. For three-step aging, combinations of high and low temperatures may be employed. According to one embodiment, heat aging treatments may be performed in accordance to MIL-H-6088. Aluminum alloy castings produced by the present  
25 method, e.g., 6000 series alloys such as 6061, are commonly heat aged after the solution heat treating step (e.g., a "T6" temperature). For example, the heat treated body may be heat aged by heating at 300-400°F, typically for about 2 to 20 hours. Aluminum alloy heat treated bodies are commonly heat aged for 5-10 hours at 325-375°F. Longer times are generally required for heat aging carried out at lower  
30 temperatures, e.g., heat aging will typically be carried out for a longer period of time at 300°F than at 400°F. The heat aging is desirably conducted for a long enough period of time to achieve desired physical properties for the cast product,

e.g., to increase the elongation of a heat treated body to at least about 6% and preferably to at least about 8%. For example, desirable cast products can be formed from 6000 series aluminum alloy (e.g., 6061) by the present method by heat aging the solution heat treated body at 325-375°F for 7-10 hours.

5        The body may undergo further mechanical or chemical processing. The exterior surface of the hipped body may be machined or "peeled" away. For example, oxides and/or other impurities may be removed from the surfaces by machining the hipped body. As the same time, machining can be used to form smooth and clean surfaces. The cast product may be rough machined to an envelope

10      slightly larger than the finished part. The inner region of unsound oxides and lower porosity is commonly removed by machining. Often the outer skin is also machined away. Parts will usually be rough machined to an envelope yielding the finished part or finish machined. Nondestructive testing (e.g., radiographic examination, fluorescent penetrate inspection, ultrasonic testing, etc.) or destructive testing (e.g.,

15      samples cut for photomicrographs) may be performed on the hipped body.

Tensile specimens of standard proportions (e.g., conforming to ASTM B 557) are generally cast with each lot of castings to size in molds representative of the practice used for the castings. Specimens may be taken from actual product castings. Metal for the specimens is part of the melt used for the castings and is

20      subjected to any grain refining additions given the metal for the castings. The temperature of the metal during pouring of the specimens should not be lower than that used during pouring of the castings.

The procedure outlined above may be used to fabricate a variety of resulting products. Such products may include, but are not limited to balls, stators, seals,

25      valve bodies, gears and large flanged bushings. Other products may include turbine and airframe components, medical equipment components, engine run components, high pressure valves and pumps, automotive parts, recreational parts that require premium surface finishes, and the like.

The process outlined above may be performed on a variety of metal alloys

30      but is particularly suitable for use with aluminum alloys. It may also find utility

with pieces cast from other metals such as cast iron, steel, stainless steel, and copper-based alloys. It is particularly suitable for use with aluminum alloy chemistries which are traditionally associated with the wrought process. For example, 6000 series wrought aluminum alloys (according to the designation of the

5      Aluminum Association in the United States) may be employed in the present method. 6000 series wrought aluminum alloys include silicon and magnesium in approximate proportions to form magnesium silicide, 6000 series alloys are generally known for being heat treatable. Alloys in the 6000 series may be formed in T4 temper or may be brought to full T6 properties by artificial heat aging.

10     According to a preferred embodiment, the 6000 series alloys include silicon and magnesium in the ratio of about 0.5:1—2:1. Mg-Si type aluminum alloys ("Al-Mg-Si-type alloys"), such as Al-Mg-Si-Cu-Cr type alloys as exemplified by 6000 series alloys, are widely used and favored for their moderately high strength, low quench sensitivity, favorable forming characteristics and corrosion resistance. In one

15     particularly suitable embodiment of the method, the 6000 series aluminum alloy is a 6061 aluminum alloy having the composition as outlined in Table 1 below:

Table 1 6061 Aluminum Alloy Composition		
Element	Minimum Wt. %	Maximum Wt. %
Magnesium	0.80	1.20
Silicon	0.40	0.80
Copper	0.15	0.40
Chromium	0.04	0.35
Iron	-----	0.70
Zinc	-----	0.25
Manganese	-----	0.15
Titanium	-----	0.15
Other Impurities, Each	-----	0.05
Other Impurities, Total	-----	0.15
Aluminum	Remainder	

Wrought 6061 aluminum alloys are used extensively in aerospace industries in different shapes and sizes. The production of cylindrical parts using wrought

techniques is generally expensive due to the process cost and acceptance standards. The present method can provide a cost effective cylindrical dense cast 6061 aluminum alloy by utilizing a combination of centrifugal casting, hot isostatic processing and heat treatment procedures. The present process can be utilized to

5 produce cast 6061-T6 aluminum alloys for lightweight simple or complex cylindrical parts requiring moderate strength and where dimensional stability is required during machining, but usage is not limited to such applications. Corrosion resistance and weldability of this alloy are generally superior to that of aluminum alloys having copper or zinc as the principle alloying element.

10 Al-Zn- type alloys, such as 7000 series wrought aluminum alloys, are another type of wrought alloy which may be employed in the present method. 7000 series alloys include zinc as the major alloying element. Other elements such as copper and chromium may be included in small quantities. 7020 and 7075 alloys are two examples of such alloys. In particular, 7075 alloys are examples of Al-Zn-

15 Mg-Cu type alloys which are suitable for use in the present method.

Al-Cu type aluminum alloys and, in particular, 2000 series wrought aluminum alloys may also be employed in the present method. 2000 series alloys include Al-Cu alloys in which copper is the principal alloying element, typically in the amount of about 2-4% by weight. Solution heat-treatment of alloys in the 2000

20 series may result in mechanical properties similar to, and which may exceed, those of mild steel. 2014, 2019, 2219, 2024 (Al-Cu-Mg-Mn type), 2124 (Al-Cu-Mg-Mn type), 2090, 2095 and 2195 are examples of suitable alloys in the 2000 series.

Al-Li type aluminum alloys and, in particular, 8000 series wrought aluminum alloys may also be utilized in the present invention. Lithium is the

25 principal alloying element in the 8000 series. 8090 is an example of a suitable Al-Zn-Mg-Cu-Cr type alloy from the 8000 series.

Traditionally cast aluminum alloys may be used in the present method. Examples of suitable cast type aluminum alloys which can be employed include 356, 319, 771, 443, 713, 336, 535, 206, 355, 850 and 851 cast aluminum alloys.

30 A cast alloy body may be produced by the method of the present invention with good physical and mechanical properties, such as high strength and/or

toughness properties. The tensile strength (i.e., a measure of the breaking stress of a material due to pulling) of an alloy body made by the present method may be in the range of about 22 - 80 KSI or higher (as determined by ASTM B 557). For example, a 6061-T6 alloy body may be produced by the present method having a  
5 tensile strength of at least about 42 KSI (290 MPa), preferably at least about 45 KSI and more preferably at least about 50 KSI. Cast bodies may be formed from 7075-T6 alloy or 2195-T8 alloy by the present method and may have a tensile strength of at least about 75 KSI or 80 KSI, respectively.

The present method may be used to produce cast aluminum bodies which  
10 exhibit good elongation and have a yield strength (i.e., the stress at which a marked and permanent increase in the deformation of a material occurs without an increase in the load; determined by ASTM B 557) in the range of about 30 to 50 KSI or higher). For example, a 6061-T6 alloy body can be produced by the present method having yield strength (2% offset) of at least about 40 KSI (275 MPa). The present  
15 method may also be used to form cast aluminum bodies from 7075-T6 alloy and 2195-T6 alloy having good elongation and tensile strength properties and 2% offset yield strengths of at least about 65 KSI and 70 KSI, respectively.

The present method can be used to produce cast aluminum alloy bodies with good tensile and yield strength and having an elongation (in 2 inches) of at least  
20 about 4%. Elongation relates to the amount a plate of the alloy bends before breaking. For example, the present method can be used to produce a 6061-T6 alloy body having an elongation of at least about 6% and, preferably, at least about 8% while still exhibiting good tensile and yield strength properties. For example, the present method permits the production of cast aluminum pieces having an elongation  
25 of 6%, a tensile strength of at least about 45 KSI and a yield strength (2% offset) of at least about 40 KSI. The present method may be used to form cast pieces from other aluminum alloys, such as 7075-T6 alloy and 2195-T6 alloy which have elongation of about 8% or higher while retaining good tensile and yield strength properties.

30 The Brinell hardness (i.e., the area of indentation produced by a hardened steel ball of 10 mm in diameter under a pressure of 500 kilograms; BHN 10/500) of an alloy body made by the present method is typically at least about 80. The

hardness is typically at least about 85 when a 10 mm ball under a pressure of 1000 kilograms (BHN 10/1000) is used to test Brinell hardness. For example, the present method permits the production of cast 6061-T6 aluminum alloy pieces having a Brinell hardness at 500 kg (BHN 10/500) in the range of about 100-120 and also

5 having tensile properties similar to those obtainable in 6061 pieces created by wrought techniques.

Examples of physical properties which can be achieved with castings produced by the present method for some exemplary wrought aluminum alloy chemistries are shown in Table 2 below:

10

Table 2  
Physical Properties of Cast Aluminum Alloys

Alloy	Tensile strength (KSI)	Yield strength (KSI)	Elongation % (2 inches)
7075-T6	75	65	8
6061-T6	48	42	8

The final cast products produced by the present method should have smooth and clean surfaces suitable for fluorescent penetrant inspection and can be subjected to fluorescent penetrant inspection of all exposed surfaces, e.g., in accordance to ASTM E 1417. Standards for acceptance are generally established by the cognizant engineering organization. Surface imperfections which can be removed so that the imperfections do not reappear on etching and do not violate the finished part envelope may be acceptable. When desired, the cast products can be subjected to ultrasonic inspection, such as in accordance with ASTM B 594. Cast pieces produced by the present method commonly meet ultrasonic Class A. The final cast products can also be subjected to radiographic examination in accordance with AMS 2635, or other acceptable technique. ASTM # 155 may be used to define radiographic acceptance standards.

The alloys and methods of the present invention may be illustrated by the following examples, which are intended to illustrate the present invention and to

teach one of ordinary skill how to make and use the invention. These examples are not intended in any way to limit or narrow the scope of protection afforded by the claims.

Example 1

5        250 Lbs. of 6061 scrap aluminum alloy was melted in a gas fired furnace where mixing of the molten metal is done manually and the temperature was brought to 1400 °F. Alloy chemistry was checked in the Lab, using spectrochemical method and additions of various elements were made as required. The chemical composition of the alloy was as follows:

Cu 0.21	Sn 0.003	Pb 0.04	Zn 0.02	Fe 0.15
Ni 0.01	Si 0.56	Mn 0.04	Mg 0.84	Ti 0.01
Cr 0.08	Al 98.02			

10       Three castings with 60 lbs. each were poured with the lot of molten aluminum into a graphite mold spinning at 700 rpm. The castings made were hipped at 960  $\pm$  25°F under an isostatic pressure of 14.750  $\pm$  250 psi for 2 hours. The hipped casting was then solution heat treated at 930°F for 6 hours and water quenched. The quenched body was then aged (precipitation hardened) at 350°F for 15 8.5 hours and finally machined for physical properties. The physical properties are as below:

<u>S/N</u>	<u>Brinell Hardness</u> (at 500 kg load (BHN))	<u>Yield Strength</u>	<u>Tensile Strength</u>	<u>Elongation %</u>
1a	71.5	24.3	36.1	9.0
1b	74.1	24.7	29.3	4.5
1c	79.6	23.6	27.9	4.25

Example 2

150 Lbs. of 6061 scrap aluminum alloy was melted in a gas fired furnace and its temperature brought to 1400 °F with manual mixing. A casting was produced using the procedure described in Example I above. Various constituent elements  
 5 were added to the melt to give the chemical composition shown below. The chemical composition and physical properties are as below:

Cu .035	Sn <0.001	Pb 0.01	Zn <0.00	Fe 0.27
Ni 0.02	Si 0.74	Mn 0.03	Mg 1.49	Ti 0.002
Cr 0.07	Al 97.01			

Physical Properties

<u>BHN At 500 kg load</u>	<u>Yield Strength</u>	<u>Tensile Strength</u>	<u>Elongation %</u>
92.6	33.27	39.11	4.0

Example 3

10 150 lbs of 6061 scrap aluminum alloy was melted in a gas fired furnace and its temperature brought to 1400 °F. One 60 Lb. casting and a test bar were produced using the procedure described in Example I above. Various constituent elements were added to the melt to give the chemical composition shown below. The chemical composition and physical properties are as below:

### Chemistry

Cu 1.9	Sn <0.0001	Pb 0.003	Zn <0.00	Fe 0.25
Ni 0.009	Si 2.36	Mn 0.036	Mg 2.19	Ti 0.1005
Cr 0.219	Al 92.86			

### Physical Properties

<u>BHN At 500 kg load</u>	<u>Yield Strength</u>	<u>Tensile Strength</u>	<u>Elongation %</u>
124	43.1	48.7	1.0

### Example 4

150 Lbs. of 6061 scrap aluminum alloy was melted in a gas fired furnace and  
5 its temperature brought to 1400 °F. One 60 Lb. casting and a test bar were produced using the procedure described in Example I above. Various constituent elements were added to the melt to give the chemical composition shown below. The chemical composition and physical properties are as below:

### Chemical Composition

Cu 2.7	Sn <0.0001	Pb 0.003	Zn <0.0001	Fe 0.25
Ni 0.007	Si 0.65	Mn 0.03	Mg 1.18	Ti 0.14
Cr 0.35	Al 94.68			

10

### Physical Properties

<u>BHN At 500 kg load</u>	<u>Yield Strength</u>	<u>Tensile Strength</u>	<u>Elongation %</u>
119	59.6	43.8	6.5

Example 5

1200 Lbs. of 6061 scrap aluminum alloy was melted in an induction furnace to have better stirring of the molten metal. Twenty 60 Lb. castings and four test bars (sample numbers denoted "TB" in Tables 3 and 4) were produced using the 5 procedure described in Example I above. Various constituent elements were added to the melt to give the chemical composition shown below. The chemical composition and physical properties of the castings are shown in Tables 3 and 4 below.

Although only a few exemplary embodiments of the present invention have 10 been described in detail in this disclosure, those skilled in the art who review this disclosure will readily appreciate that many modifications are possible in the exemplary embodiments (such as variations in sizes, structures, shapes and proportions of the various elements, values of parameters, mounting arrangements, or use of materials) without materially departing from the novel teachings and 15 advantages of the invention.

TABLE 3

Sample No.	Cu	Sn	Pb	Zn	Fe	Ni	Si	Mn	Mg	Ti	Cr	Base
1a/TB	0.3248	<0.0001	0.0004	<0.0001	0.2191	0.0043	0.6775	0.0236	0.9049	0.2065	0.0164	97.562
1b/TB	0.3108	<0.0001	0.0030	<0.0001	0.2278	0.0037	0.6542	0.0235	0.9087	0.0605	0.2291	97.579
1c/TB	0.3108	<0.0001	0.0030	<0.0001	0.2278	0.0037	0.6542	0.0235	0.9087	0.0605	0.2291	97.579
1a	0.2911	<0.0001	0.0018	<0.0001	0.2611	0.0049	0.7018	0.0252	1.018	0.0382	0.2309	97.428
1b	0.3078	<0.0001	<0.0001	<0.0001	0.2478	0.0051	0.5918	0.0249	1.013	0.0147	0.2151	97.479
2a	0.3963	<0.0001	0.0032	<0.0001	0.2580	0.0068	0.6721	0.0221	0.9148	0.0754	0.2593	97.388
2b	0.2980	<0.0001	0.0014	<0.0001	0.2120	0.0036	0.6078	0.0230	0.8704	0.0708	0.2505	97.654
3a	0.4107	<0.0001	0.0094	<2.0001	0.2346	0.0061	0.6550	0.0240	0.9605	0.0494	0.2374	97.413
3b	0.3327	<0.0001	0.0016	<0.0001	0.2207	0.0046	0.6528	0.0233	0.9558	0.0361	0.2195	97.553
4a	0.3293	<0.0001	0.0064	<0.0001	0.2461	0.0038	0.6534	0.0222	0.8672	0.0273	0.2130	97.631
4b	0.3268	<0.0001	0.0017	<0.0001	0.2292	0.0041	0.6503	0.0236	0.9550	0.0529	0.2333	97.523
5a	0.3336	<0.0001	0.0017	<0.0001	0.2313	0.0047	0.6609	0.0242	0.9692	0.0215	0.1994	97.553
5b	0.3449	<0.0001	0.0001	<0.0001	0.2330	0.0048	0.6535	0.0244	0.9932	0.0215	0.2079	97.517

Sample No.	Cu	Sn	Pb	Zn	Fe	Ni	Si	Mn	Mg	Ti	Cr	Base
2a/TB	0.2817	<0.0001	0.0015	<0.0001	0.2018	0.0034	0.5671	0.0231	1.004	0.1797	0.3178	97.42
2b/TB	0.3127	<0.0001	0.0012	<0.0001	0.2276	0.0034	0.6040	0.0238	1.043	0.1605	0.3169	97.307
6a	0.4227	<0.0001	0.0041	<0.0001	0.2327	0.0057	0.6756	0.0238	1.118	0.0730	0.2970	97.147
6b	0.3893	<0.0001	0.0021	<0.0001	0.2369	0.0052	0.6805	0.0238	1.121	0.0646	0.2964	97.180
7a	0.3085	<0.0001	<0.0001	<0.0001	0.2291	0.0038	0.6165	0.0237	1.038	0.1387	0.3099	97.332
7b	0.3458	<0.0001	0.0032	<0.0001	0.2376	0.0048	0.6127	0.0245	1.023	0.1385	0.3191	97.290
8a	0.3316	<0.0001	0.0011	<0.0001	0.2231	0.0042	0.6249	0.0237	1.040	0.0749	0.3031	97.393
8b	0.3321	<0.0001	0.0018	<0.0001	0.2081	0.0042	0.6151	0.0233	1.024	0.0913	0.3056	97.394
9a	0.3267	<0.0001	0.0001	<0.0001	0.2149	0.0046	0.6050	0.0224	0.9811	0.0964	0.3158	97.433
9b	0.3410	<0.0001	0.0016	<0.0001	0.2280	0.0049	0.6247	0.0231	1.002	0.1269	0.3152	97.333
10a	0.3360	<0.0001	0.0015	<0.0001	0.2148	0.0039	0.6162	0.0232	1.008	0.0795	0.3071	97.410
10b	0.3380	<0.0001	0.0007	<0.0001	0.1978	0.0044	0.6126	0.0224	0.9916	0.0648	0.3110	97.451
3a/TB	0.3527	<0.0001	0.0016	<0.0001	0.2343	0.0041	0.6478	0.0205	0.9912	0.1530	0.2995	97.296
3b/TB	0.3720	<0.0001	0.0013	<0.0001	0.2356	0.0055	0.6439	0.0213	0.9924	0.1387	0.3071	97.262

Sample No.	Cu	Sn	Pb	Zn	Fe	Ni	Si	Mn	Mg	Ti	Cr	Base
11a	0.3044	<0.0001	0.0055	<0.0001	0.2151	0.0040	0.6002	0.0193	0.9204	0.1851	0.2988	97.447
11b	0.3119	<0.0001	0.0026	0.2231	0.0041	0.5946	0.0193	0.8978	0.1878	0.1882	0.3023	97.456
12a	0.4455	<0.0001	0.0095	<0.0001	0.2560	0.0059	0.6747	0.0204	1.001	0.2024	0.3069	97.078
12b	0.3594	<0.0001	0.0008	<0.0001	0.2278	0.0039	0.6681	0.0207	1.042	0.1316	0.2912	97.254
13a	0.2709	<0.0001	0.0028	<0.0001	0.2353	0.0028	0.5919	0.0177	0.8611	0.2164	0.2967	97.534
13b	0.3033	<0.0001	0.0011	<0.0001	0.2145	0.0037	0.5952	0.0191	0.8916	0.2309	0.3062	97.445
14a	0.2957	<0.0001	0.0019	<0.0001	0.2190	0.0040	0.5999	0.0191	0.8948	0.2073	0.2985	97.460
14b	0.2931	<0.0001	0.0010	<0.0001	0.2173	0.0041	0.5794	0.0192	0.8807	0.2094	0.3099	97.497
15a	0.3020	<0.0001	0.0025	<0.0001	0.2027	0.0037	0.6202	0.0183	0.9182	0.1737	0.2845	97.474
15b	0.3298	<0.0001	0.0074	<0.0001	0.2059	0.0047	0.6047	0.0188	0.8934	0.1939	0.2980	97.451
4a/TB	0.1984	<0.0001	0.0029	<0.0001	0.2218	0.0034	0.6382	0.0198	0.9188	0.1240	0.1984	97.674
4b/TB	0.2077	<0.0001	0.0021	<0.0001	0.2307	0.0040	0.6303	0.0199	0.9049	0.1290	0.2009	97.670
4c/TB	0.2752	<0.0001	0.0048	<0.0001	0.2334	0.0053	0.6267	0.0198	0.8916	0.1515	0.2039	97.588
16a	0.2278	<0.0001	0.0053	<0.0001	0.2288	0.0054	0.6136	0.0195	0.8602	0.1533	0.2042	97.682

Sample No.	Cu	Sn	Pb	Zn	Fe	Ni	Si	Mn	Mg	Ti	Cr	Base
16b	0.2278	<0.0001	0.0053	<0.0001	0.2288	0.0054	0.6136	0.0195	0.8602	0.1533	0.2042	97.682
17a	0.2601	<0.0001	0.0041	<0.0001	0.2243	0.0064	0.6327	0.0186	0.8987	0.1451	0.1974	97.613
17b	0.2010	<0.0001	0.0026	<0.0001	0.2367	0.0034	0.6325	0.0194	0.9179	0.1236	0.1944	97.668
18a	0.2102	<0.0001	0.0094	<0.0001	0.2261	0.0049	0.5937	0.0196	0.8362	0.1706	0.2072	97.722
18b	0.2226	<0.0001	0.0057	<0.0001	0.2281	0.0059	0.5802	0.0199	0.8342	0.1747	0.2091	97.720
19a	0.2176	<0.0001	0.0021	<0.0001	0.2420	0.0036	0.6676	0.0193	0.9396	0.1064	0.1896	97.612
19b	0.1752	<0.0001	0.0021	<0.0001	0.2259	0.0043	0.5901	0.0188	0.8983	0.1586	0.1991	97.778
20a	0.3805	<0.0001	0.0185	<0.0001	0.2262	0.0082	0.6254	0.0183	0.8524	0.1600	0.2024	97.508
20b	0.1803	<0.0001	0.0018	<0.0001	0.2270	0.10042	0.5950	0.0188	0.8303	0.1515	0.2000	97.791

TABLE 4

Sample No.	Yield Strength	Tensile Strength	% Elongation	Comments	Hardness BHN at 500g
1a/TB	45.69	49.17	3.0		109
1b/TB	45.58	49.13	3.0		109
1c/TB	49.40	50.0	4.0		109
1a	42.51	45.59	2.5	Not hipped	109
1b	42.51	45.56	2.5	Not hipped	109
2a	45.73	50.89	6.0		109
2b	45.64	50.45	6.0		109
3a	44.42	48.95	4.0		109
3b	44.23	48.83	4.0		109
4a	44.37	48.80	4.0		109
4b	44.27	48.86	4.0		109
5a	44.39	48.96	3.5		109
5b	44.30	48.81	3.5		109
2a/TB	42.91	49.49	6.5		109
2b/TB	43.02	49.07	6.5		109
6a	40.81	43.40	2.5	Not Hipped	109
6b	40.73	42.37	1.5	Not Hipped	109
7a	42.74	48.65	5.0		109
7b	42.79	48.63	5.5		109
8a	42.32	48.42	6.0		109
8b	42.14	48.34	5.5		109
9a	43.17	47.73	3.5		109
9b	43.23	47.79	4.0		109
10a	43.25	48.19	4.5		109
10b	43.36	48.36	4.5		109

Sample No.	Yield Strength	Tensile Strength	% Elongation	Comments	Hardness BHN at 500g
3a/TB	44.98	50.47	5.0		109
3b/TB	45.02	50.41	5.5		109
11a	42.33	45.59	2.0	Not Hipped	109
11b	42.19	45.48	2.0	Not Hipped	109
12a	44.24	49.30	4.5		109
12b	44.43	49.28	4.0		109
13a	44.39	49.19	4.0		109
13b	44.52	49.19	4.0		109
14a	45.22	49.50	4.0		109
14b	45.23	49.59	4.0		109
15a	44.89	49.15	3.5		109
15b	44.95	49.12	3.5		109
4a/TB	44.54	45.98	1.5		109
4b/TB	44.74	49.92	6.0		109
4c/TB	41.40	51.50	8.0		109
16a	42.67	45.41	2.0	Not Hipped	109
16b	42.54	45.26	2.0	Not Hipped	109
17a	44.77	48.44	3.5		109
17b	44.57	48.55	3.5		109
18a	44.95	48.18	3.5		109
18b	44.87	48.25	3.0		109
19a	44.77	48.33	3.0		109
19b	44.75	48.14	3.0		109
20a	44.44	47.71	3.0		109
20b	44.68	47.89	2.5		109

WHAT IS CLAIMED IS:

1. A method for producing a high strength cast aluminum alloy body comprising:
  - providing a molten aluminum alloy;
  - centrifugally casting the molten aluminum alloy in a mold to form a cast body; and
  - hot isostatically processing the cast body to form a hipped body.
2. The method of 1, further comprising solution heat treating the hipped body to form a heat treated body.
3. The method of claim 2, wherein the solution heat treating step comprises heating the hipped body to at least about 850°F.
4. The method of claim 3, wherein the solution heat treating step comprises heating the hipped body for at least about 2 hours.
5. The method of claim 4, wherein the aluminum alloy is a 6061 aluminum alloy and the solution heat treating step comprises heating the hipped body to about 900 to 950°F for about 4 to 10 hours.
6. The method of claim 2, further comprising heat aging the heat treated body to form an aged body.
7. The method of claim 6, wherein the heat aging step comprises heating the heat treated body at about 300 to 400°F.
8. The method of claim 6, wherein the heat aging step comprises heating the heat treated body for about 2 to 20 hours.
9. The method of claim 7, wherein the aluminum alloy is a 6061 aluminum alloy and the heat aging step comprises heating the heat treated body at about 325 to 375°F for about 7 to 10 hours.

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10. The method of claim 6, further comprising machining the aged body to remove an impurity region.
11. The method of claim 1, wherein the centrifugally casting step comprises rotating a mold at a speed of at least about 500 rpm.
12. The method of claim 1, wherein the centrifugally casting step comprises centrifugally casting the molten aluminum alloy at a centrifugal acceleration of at least about 30 G.
13. The method of 2, further comprising machining the heat treated body to remove an impurity region.
14. The method of claim 1, wherein the hot isostatically processing step comprises heating the cast body at a temperature of at least about 900°F while applying an isostatic pressure of at least about 10 KSI.
15. The method of claim 14, wherein the hot isostatically processing step comprises heating the cast body to a temperature of about 935 to 985°F for at least one hour while applying an isostatic pressure of at least about 14 KSI.
16. The method of claim 1, wherein the step of providing a molten aluminum alloy comprises forming melted aluminum alloy in an induction furnace.
17. A high strength cast aluminum alloy product produced by a process comprising the steps of:
  - providing a molten body of a melted aluminum alloy;
  - centrifugally casting the molten body to form a cast body; and
  - hot isostatically processing the cast body to form a hipped body.
18. An article formed from an aluminum alloy having a generally round grain structure and being substantially free of microshrinkage defects.
19. The article of claim 18, wherein the aluminum alloy has an average grain size is about 3000 to 4000  $\mu$ inch.

20. The article of claim 18, wherein the aluminum alloy is a wrought aluminum alloy having sufficient fluidity as a melt for centrifugal casting.

21. The article of claim 20, wherein the wrought aluminum alloy is selected from the group consisting of series 2000, 4000, 6000, 7000 and 8000 series aluminum alloys.

22. The article of claim 20, wherein the wrought alloy is a 2024, 2090, 2095, 2195 or 2219 aluminum alloy.

23. The article of claim 20, wherein the wrought alloy is a 6061 aluminum alloy.

24. The article of claim 23, wherein the wrought alloy is a 6061-T6 aluminum alloy.

25. The article of claim 20, wherein the wrought alloy is a 7075 aluminum alloy.

26. The article of claim 20, wherein the wrought alloy is a 8090 aluminum alloy.

27. The article of claim 18, wherein the aluminum alloy is an Al-Mg-Si type aluminum alloy.

28. The article of claim 26, wherein the aluminum alloy comprises 0.4-0.8% Si, 0.15-0.4% Cu, 0.04-0.35% Cr, 0.8-1.2% Mg, 0.05-0.7% Fe and at least 94.85 wt % Al.

**29.** The article of claim 18, wherein the aluminum alloy is an Al-Cu type aluminum alloy.

**30.** The article of claim 18, wherein the aluminum alloy is an Al-Zn type aluminum alloy.

31. The article of claim 18, wherein aluminum alloy has an elongation at least about 4%.

32. The article of claim 18, wherein the aluminum alloy article has a tensile strength, a yield strength and an elongation meeting ASTM wrought specifications.

33. A cast aluminum alloy article formed from a 6000 series aluminum alloy and having an elongation of at least about 4% and a tensile strength of at least about 38 KSI.

34. The article of claim 33 having a 0.2 % offset yield strength of at least about 32 KSI.

35. The article of claim 34 having a tensile strength of at least about 50 KSI.

36. The article of claim 34 having an elongation of at least 8%.

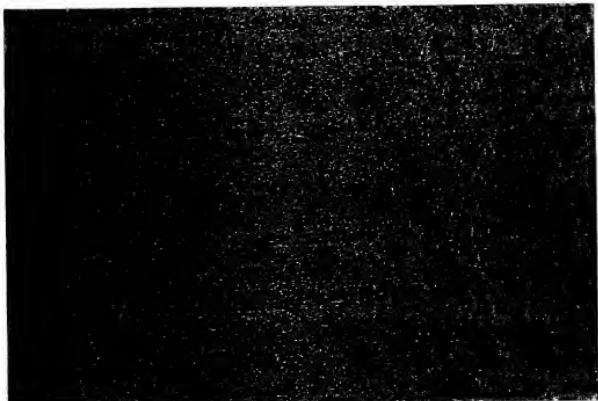
37. The article of claim 34 having a 0.2 % offset yield strength of at least about 45 KSI.

38. The article of claim 33 having an elongation of at least 6%, a tensile of at least about 45 KSI and a 0.2 % offset yield strength of at least about 40 KSI.

39. The article of claim 33 having a Brinell Hardness at 500 kg load of at least about 80.

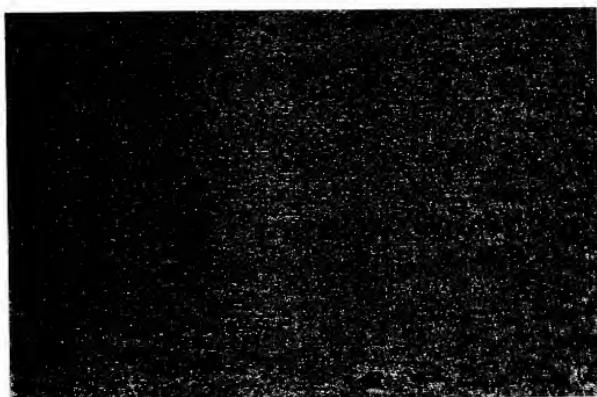
## **ABSTRACT**

A method for producing a cast aluminum alloy article having high strength and/or toughness is provided. The method includes providing a molten aluminum alloy, centrifugally casting the molten aluminum alloy to form a cast body; and hot isostatically processing the cast body to form a hipped body. The hipped body may optionally be solution heat treated to form a heat treated body, which may subsequently be precipitation hardened to further enhance the properties of the cast product as desired. The method allows the production of cast aluminum alloy articles having physical and mechanical properties similar to those obtained for articles produced from corresponding aluminum alloy chemistries by wrought techniques.



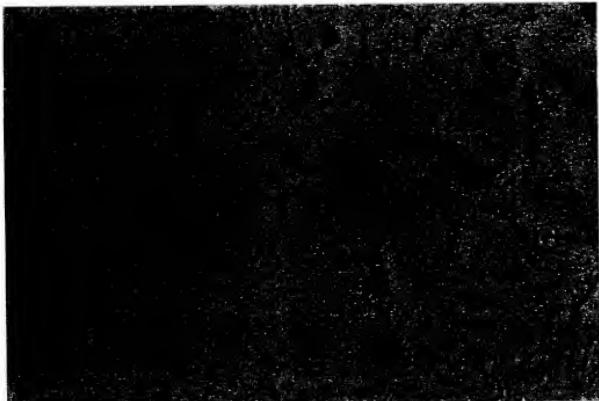
### **200X 6061-T651 Wrought Plate**

Fig 2



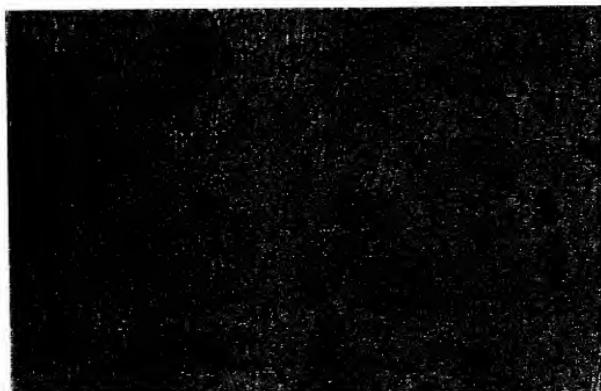
### **100X 6061-T651 Wrought Plate**

FIG. 1



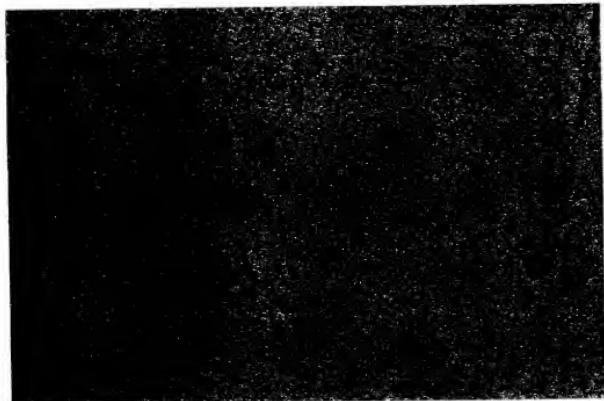
200X Unhipped Centrifugal Cast JC6061-T6C

FIG. 4



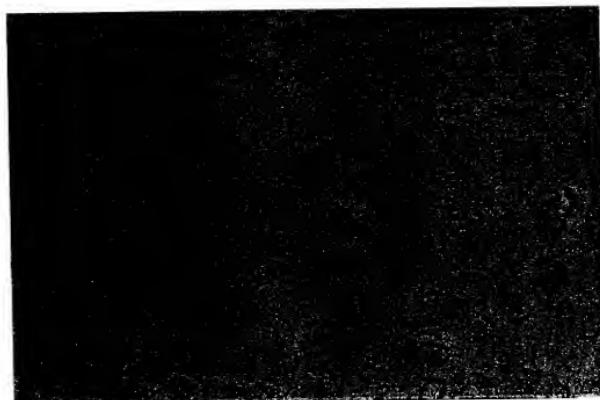
100X Unhipped Centrifugal Cast JC6061-T6C

FIG. 3



50X Centrifugal Cast and Hipped JC6061-T6C

FIG. 5



100X Centrifugal Cast and Hipped JC6061-T6C

FIG. 6



200X Centrifugal Cast and Hipped JC6061-T6C

FIG. 7

**DECLARATION AND POWER OF ATTORNEY**

As a below named inventor, I HEREBY DECLARE:

THAT my residence, post office address, and citizenship are as stated below next to my name;

THAT I believe I am the original, first, and sole inventor (if only one inventor is named below) or an original, first, and joint inventor (if plural inventors are named below or in an attached Declaration) of the subject matter which is claimed and for which a patent is sought on the invention entitled

**METAL ALLOY PRODUCT AND METHOD FOR PRODUCING SAME**

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(Attorney Docket No. 43420-117)

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the specification of which (check one)

is attached hereto.

X

was filed on August 17, 1999 as United States Application Number or PCT International Application Number 09/376,067 and was amended on \_\_\_\_\_ (if applicable).

THAT I do not know and do not believe that the same invention was ever known or used by others in the United States of America, or was patented or described in any printed publication in any country, before I (we) invented it;

THAT I do not know and do not believe that the same invention was patented or described in any printed publication in any country, or in public use or on sale in the United States of America, for more than one year prior to the filing date of this United States application;

THAT I do not know and do not believe that the same invention was first patented or made the subject of an inventor's certificate that issued in any country foreign to the United States of America before the filing date of this United States application if the foreign application was filed by me (us), or by my (our) legal representatives or assigns, more than twelve months (six months for design patents) prior to the filing date of this United States application;

THAT I have reviewed and understand the contents of the above-identified specification, including the claim(s), as amended by any amendment specifically referred to above;

THAT I believe that the above-identified specification contains a written description of the invention, and of the manner and process of making and using it, in such full, clear, concise, and exact terms as to enable any person skilled in the art to which it pertains, or with which it is most nearly connected, to make and use the invention, and sets forth the best mode contemplated by me of carrying out the invention; and

THAT I acknowledge the duty to disclose to the U.S. Patent and Trademark Office all information known to me to be material to patentability as defined in Title 37, Code of Federal Regulations, §1.56.

I HEREBY CLAIM foreign priority benefits under Title 35, United States Code §119(a)-(d) or § 365(b) of any foreign application(s) for patent or inventor's certificate, or §365(a) of any PCT international application which designated at least one country other than the United States of America, listed below and have also identified below any foreign application for patent or inventor's certificate or of any PCT international application having a filing date before that of the application on which priority is claimed.

Prior Foreign Application Number	Country	Foreign Filing Date	Priority Claimed?	Certified Copy Attached?

I HEREBY CLAIM the benefit under Title 35, United States Code § 119(e) of any United States provisional application(s) listed below.

U.S. Provisional Application Number	Filing Date

I HEREBY CLAIM the benefit under Title 35, United States Code, §120 of any United States application(s), or § 365(c) of any PCT international application designating the United States of America, listed below and, insofar as the subject matter of each of the claims of this application is not disclosed in the prior United States or PCT International application in the manner provided by the first paragraph of Title 35, United States Code, § 112, I acknowledge the duty to disclose information which is material to patentability as defined in Title 37, Code of Federal Regulations, § 1.56 which became available between the filing date of the prior application and the national or PCT international filing date of this application.

U.S. Parent Application Number	PCT Parent Application Number	Parent Filing Date	Parent Patent Number

I HEREBY APPOINT the following registered attorneys and agents of the law firm of FOLEY & LARDNER to have full power to prosecute this application and any continuations, divisions, reissues, and reexaminations thereof, to receive the patent, and to transact all business in the United States Patent and Trademark Office connected therewith:

RUSSELL J. BARRON	Reg. No. 29,512
DAVID J. BATES	Reg. No. 39,902
STEVEN C. BECKER	Reg. No. 42,308
DOUGLAS A. BOEHM	Reg. No. 32,014
EDWARD W. BROWN	Reg. No. 22,022
LISA A. BRZYCKI	Reg. No. 40,926
CHARLES G. CARTER	Reg. No. 35,093
JOHN C. COOPER III	Reg. No. 26,416
JEFFREY N. COSTAKOS	Reg. No. 34,144
WILLIAM J. DICK	Reg. No. 22,205

BARRY L. GROSSMAN	Reg. No. 30,844
KEITH D. LINDENBAUM	Reg. No. 40,365
RICHARD J. MC KENNA	Reg. No. 35,610
JAMES G. MORROW	Reg. No. 32,505
S. MICHAEL PATTON	Reg. No. 36,235
JOHN T. PIENKOS	Reg. No. 42,997
TODD A. RATHE	Reg. No. 38,276
CHRISTOPHER M. TUROSKI	Reg. No. 44,456
PETER J. VOGEL	Reg. No. 41,363
JAMES A. WILKE	Reg. No. 34,279
JOSEPH N. ZIEBERT	Reg. No. 35,421
WALTER E. ZIMMERMAN	Reg. No. 40,883

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and I request that all correspondence be directed to:

Charles G. Carter  
 FOLEY & LARDNER  
 Firstar Center  
 777 East Wisconsin Avenue  
 Milwaukee, Wisconsin 53202-5367  
 Telephone: (414) 297-5842  
 Facsimile: (414) 297-4900

I UNDERSTAND AND AGREE THAT the foregoing attorneys and agents appointed by me to prosecute this application do not personally represent me or my legal interests, but instead represent the interests of the legal owner(s) of the invention described in this application.

I FURTHER DECLARE THAT all statements made herein of my own knowledge are true, and that all statements made on information and belief are believed to be true; and further that these statements were made with the knowledge that willful false statements and the like so made are punishable by fine or imprisonment, or both, under Section 1001 of Title 18 of the United States Code, and that such willful false statements may jeopardize the validity of the application or any patent issuing thereon.

Name of first inventor	Steven A. Clark
Residence	Hales Corners, WI
Citizenship	United States of America
Post Office Address	5621 S. 106th Street, Hales Corners, WI 53130
Inventor's signature	<i>Steven A. Clark</i>
Date	1/4/99

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Name of second inventor

Balathandan S. Pillai

Residence

Glendale, WI

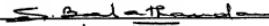
Citizenship

India

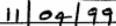
Post Office Address

921 W. Theresa Lane, Glendale, WI 53209

Inventor's signature



Date

  
11/04/99

11/12/2010 9:23:52 AM

***IN THE UNITED STATES PATENT AND TRADEMARK OFFICE***

Applicant: Steven A. Clark et al.

Title: METAL ALLOY PRODUCT AND  
METHOD FOR PRODUCING  
SAME

Appl. No.: 09/376,067

Filing Date: 8/17/1999

Examiner: To be determined

Art Unit: 1742

**POWER OF ATTORNEY OR AUTHORIZATION OF AGENT  
BY ASSIGNEE**

Assistant Commissioner for Patents  
Washington, D.C. 20231

Sir:

Johnson Brass & Machine Foundry, Inc., a corporation duly organized and existing under the laws of the State of Wisconsin, and having a place of business at 270 North Mill Street, P.O. Box 219, Saukville, Wisconsin 53080 is the Assignee and owner of the entire right, title, and interest in and to the above-captioned patent application.

Assignee hereby appoints the following registered attorneys and agents of the law firm of Foley & Lardner to have full power to prosecute this application and any continuations, divisions, reissues, and reexaminations thereof, to receive the patent, and to transact all business in the United States Patent and Trademark Office connected therewith:

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JAMES G. MORROW	Reg. No. 32,505
RICHARD B. O'PLANICK	Reg. No. 29,096

STEPHEN MICHAEL PATTON	Reg. No. 36,235
JOHN T. PIENKOS	Reg. No. 42,997
TODD A. RATHE	Reg. No. 38,276
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CHRISTOPHER M. TUROSKI	Reg. No. 44,456
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JAMES A. WILKE	Reg. No. 34,279
JOSEPH N. ZIEBERT	Reg. No. 35,421
WALTER E. ZIMMERMAN	Reg. No. 40,883

and requests that all correspondence be directed to:

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Telephone: (414) 297-5842  
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Executed this 1 day of November, 1999.

Johnson Brass & Machine Foundry, Inc.

By:

Lance Johnson

(Signature)

Lance Johnson

(Printed Name)

President

(Title)

United States Patent & Trademark Office  
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Drawings